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	Numerical m	odeling of sul	omarine mass	-movemen	t generated					
	waves using RANS model									
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		Received 31 October 2003; accepted 26 October 2005								
	Abstract			2						
	In this paper a numerical	model for predicting way	ves generated by nearshor	e submarine mass-	movements is described.					

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The model is based on the Reynolds averaged Navier–Stokes (RANS) equations with the  $k-\varepsilon$  turbulence model. The 23 volume of fluid (VOF) method is employed to track the free surface. Numerical results obtained from the present model are validated with laboratory experiments and analytical solutions. Very good agreements are observed for both submarine 25 and aerial mass movements. Numerical experiments are performed to obtain the empirical formula for the maximum

runup and rundown as functions of slide properties.

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Keywords: Submarine mass movement; Numerical model; Turbulence; Breaking waves 29

#### 31

#### 1. Introduction

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Motivated by the needs for preservation of 35 human lives and coastal infrastructure, and for the deployment and operation of special structural and 37

mechanical systems in coastal areas, the study of nearshore wave motions and wave-structure inter-

39 action has been of interest to coastal scientists and engineers for many years.

41 Coastal wave generation due to submarine mass movement is a complex process. While the length

scale of a submarine mass movement is usually 43 smaller than that of a seafloor displacement created

45 by a fault rupture, the time-scale is usually longer. Therefore, the concept of "initial free surface 47

displacement" in the wave generation region becomes a critical issue. Hence the evolution of the 55 free surface displacement in the source region of mass movement needs to be modeled entirely. 57 Furthermore, the characteristics of a submarine mass movement, including the soil properties, 59 volume and area of the mass movement, also require a post-event bathymetry survey. 61

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Several numerical models have been developed to describe the waves generated by submerged or aerial 63 mass movements. With the common assumption that the geometry and the time history of the mass 65 movement can be prescribed, these models adopt various additional approximations in hydrody-67 namics. For instance, Lynett and Liu (2002) presented a model based on the depth-integrated 69 nonlinear wave equations, which include the frequency dispersion effects. Therefore, their model 71

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1 can simulate relatively short waves that might be generated by a submarine mass movement. Grilli and Watts (1999) adopted a boundary integral 3 equation method (BIEM), based on the potential 5 flow theory, and developed a fully nonlinear model for mass movement-generated waves. However, the

7 approach does not take into account wave breaking, which could be important in the vicinity of the 9 generation region as well as the runup region. The

depth-averaged model suffers the same drawback as 11 the BIEM model in terms of the lack of capability of

modeling breaking waves. However, it is much more 13 computationally efficient as it has reduced the 3D

problem to a 2D problem in the horizontal space. 15 Heinrich (1992) modified the NASA-VOF2D model, which is a 2D (vertical plane) nonlinear free

17 surface model based on the Navier-Stokes equations, to study the generation, propagation and

19 runup of tsunamis created by landslides. The effects of turbulence are not considered. Heinrich com-

21 pared his numerical results for both submarine and aerial mass movements with his own experiments.

23 The agreement is reasonable, except in the regions where wave-breaking-induced turbulence is impor-25 tant.

In recent years, significant advancement in 27 modeling wave-breaking process and interactions between breaking waves and coastal structures has 29 been made. For example, Cornell breaking waves and structures model (COBRAS) is based on the 31 Reynolds Averaged Navier-Stokes (RANS) equations with a  $k-\varepsilon$  turbulence closure model. While a 33 nonlinear Reynolds stress model is employed to allow anisotropic turbulence, the volume of fluid 35 (VOF) method is used to track the free surface movements. COBRAS has been verified and vali-37 dated by comparing numerical results with experimental data for runup and rundown of breaking

39 waves on a uniform beach (Lin and Liu, 1998a,b; Lin et al., 1999). It also has the capability of 41 simulating wave-structure interactions, where the structures are rigid, stationary, fully submerged or

43 surface piercing (Hsu et al., 2002). The primary goal of this paper is to modify

45 COBRAS to allow time-dependent moving solid boundaries such that mass movement-created waves 47 can be simulated. Since COBRAS is capable of

calculating turbulence, the modified model will be 49 able to simulate breaking waves, runup and run-

down. Here, we shall first present briefly the 51 theoretical background of COBRAS and discuss the necessary modification to simulate the mass movement. 2D numerical results are then compared 53 with experimental data. Some discussions on the 55 future extensions are given at the end of the paper.

#### 57 2. Description of the model

59 In this section the mathematical formulation and the associated numerical algorithm of COBRAS are 61 discussed briefly. More detailed discussions can be found in Lin and Liu (1998 a, b). The model is based 63 on the RANS equations. For a turbulent flow, the velocity field and pressure field can be decomposed 65 into two parts: the mean (ensemble average) velocity and pressure  $\langle u_i \rangle$  and  $\langle p \rangle$ , and the deviatoric (or 67 turbulent) velocity and pressure  $u'_i$  and p'. Thus,  $u_i = \langle u_i \rangle + u'_i$  and  $p = \langle p \rangle + p'$  in which i = 1, 2, 369 for a 3D flow. If the fluid is assumed incompressible, the mean flow field is governed by the RANs 71 equations:

$$\frac{\partial \langle u_i \rangle}{\partial x_i} = 0, \tag{1} 73$$

$$\frac{\partial \langle u_i \rangle}{\partial t} + \langle u_j \rangle \frac{\partial \langle u_i \rangle}{\partial x_j} = -\frac{1}{\rho} \frac{\partial \langle p \rangle}{\partial x_i} + g_i + \frac{1}{\rho} \frac{\partial \langle \tau_{ij} \rangle}{\partial x_j} - \frac{\partial \langle u'_i u'_j \rangle}{\partial x_j}$$
(2)

79 in which  $\rho$  is the density of the fluid,  $g_i$  the *i*th component of the gravitational acceleration, and 81 the mean molecular stress tensor  $\langle \tau_{ij} \rangle = 2\mu \langle \sigma_{ij} \rangle$  with  $\mu$ , the molecular viscosity and  $\langle \sigma_{ii} \rangle$ , the rate of strain 83 tensor of the mean flow. In the momentum equation (2), the influence of the turbulent fluctuations on the 85 mean flow field is represented by the Reynolds stress tensor  $-\rho \langle u'_i u'_i \rangle$ . Many second-order turbulence 87 closure models have been developed for different applications. In the present model, the Reynolds 89 stress is approximated by a nonlinear algebraic stress model:

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$$\rho \langle u'_i u'_j \rangle = \frac{2}{3} \rho k \delta_{ij} - C_d \frac{k^2}{\varepsilon} \left( \frac{\partial \langle u_i \rangle}{\partial x_j} + \frac{\partial \langle u_j \rangle}{\partial x_i} \right)$$
93

$$\begin{bmatrix} C_1 \begin{pmatrix} \frac{\partial\langle u_i \rangle}{\partial x_l} \frac{\partial\langle u_l \rangle}{\partial x_j} + \frac{\partial\langle u_i \rangle}{\partial x_l} \frac{\partial\langle u_l \rangle}{\partial x_i} \\ -\frac{2}{2} \frac{\partial\langle u_l \rangle}{\partial x_i} \frac{\partial\langle u_k \rangle}{\partial x_i} & \end{bmatrix} \qquad 95$$

$$-\rho \frac{k^3}{\varepsilon^2} \begin{vmatrix} 3 & 3 & 3 & 0 \\ +C_2 \left(\frac{\partial \langle u_i \rangle}{\partial x_i} \frac{\partial \langle u_j \rangle}{\partial x_i} - \frac{1}{3} \frac{\partial \langle u_i \rangle}{\partial x_i} \frac{\partial \langle u_i \rangle}{\partial x_i} \delta_{ij} \right) \end{vmatrix} \qquad 97$$

$$+C_{3}\left(\frac{\partial\langle u_{k}\rangle}{\partial x_{i}}\frac{\partial\langle u_{k}\rangle}{\partial x_{j}}-\frac{1}{3}\frac{\partial\langle u_{l}\rangle}{\partial x_{k}}\frac{\partial\langle u_{l}\rangle}{\partial x_{k}}\delta_{ij}\right)\right] 101$$
(3)

in which  $C_d, C_1, C_2$  and  $C_3$  are empirical coeffi-

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#### D. Yuk et al. / Computers & Geosciences I (IIII) III-III

- 1 cients,  $\delta_{ij}$  the Kronecker delta,  $k = \langle u'_i u'_i \rangle/2$  the turbulent kinetic energy, and  $\varepsilon = v \langle (\partial u'_i / \partial x_i)^2 \rangle$  the
- dissipation rate of turbulent kinetic energy, where 3  $v = \mu/\rho$  is the molecular kinematic viscosity. It is 5 noted that for the conventional eddy viscosity model  $C_1 = C_2 = C_3 = 0$  in (3) and the eddy viscosity is then expressed as  $v_t = C_d k^2 / \varepsilon$ . Com-7 pared with the conventional eddy viscosity model,
- 9 the nonlinear Reynolds stress model (3) can be applied to general anisotropic turbulent flows.
- The governing equations for k and  $\varepsilon$  are modeled 11 as (Lin and Liu, 1998a, b)

$$\frac{13}{15} \quad \frac{\partial k}{\partial t} + \langle u_j \rangle \frac{\partial k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[ \left( \frac{v_t}{\sigma_k} + v \right) \frac{\partial k}{\partial x_j} \right] 
- \langle u'_i u'_j \rangle \frac{\partial \langle u_i \rangle}{\partial x_j} - \varepsilon,$$
(4)

19  $\frac{\partial \varepsilon}{\partial t} + \langle u_j \rangle \frac{\partial \varepsilon}{\partial x_i}$  $= \frac{\partial}{\partial x_i} \left[ \left( \frac{v_t}{\sigma_k} + v \right) \frac{\partial \varepsilon}{\partial x_i} \right]$ 21 23  $+ C_{1\varepsilon} \frac{\varepsilon}{k} v_t \left( \frac{\partial \langle u_i \rangle}{\partial x_i} + \frac{\partial \langle u_j \rangle}{\partial x} \right) \frac{\partial \langle u_i \rangle}{\partial x_i} - C_{2\varepsilon} \frac{\varepsilon^2}{k}$ (5)

in which  $\sigma_{k}, \sigma_{\varepsilon}, C_{1\varepsilon}$  and  $C_{2\varepsilon}$ , are empirical coeffi-27 cients. The coefficients in Eqs. (3)-(5) have been determined by performing many simple experiments 29 and enforcing the physical realizability; the recommended values for these coefficients can be found in 31 Lin and Liu (1998a, b).

- Appropriate boundary conditions need to be specified. For the mean flow field, both the no-slip 33 and the free-slip boundary condition can be 35 imposed on the solid boundary. Along the mass surface, the velocity of the moving boundary is 37 either prescribed or determined by dynamic equili-
- brium of the mass. The zero-stress condition is 39 required on the mean free surface by neglecting the effect of airflow. For the turbulent field, near the
- 41 solid boundary, the log-law distribution of mean tangential velocity in the turbulent boundary layer
- 43 is applied so that the values of k and  $\varepsilon$  can be expressed as functions of distance from the bound-45
- ary and the mean tangential velocity outside the viscous sublayer. On the free surface, the zero-47 gradient boundary conditions are imposed for both
- k and  $\varepsilon$ , i.e.,  $\partial k/\partial n = \partial \varepsilon/\partial n = 0$ . A low level of k for 49 the initial and inflow boundary conditions is
- assumed. 51 In the numerical model, the RANS equations are solved by a finite difference two-step projection

method. The forward time difference method is used 53 to discretize the time derivative. The advection terms are discretized by the combination of central 55 difference method and upwind method. The central difference method is employed to discretize the 57 pressure gradient terms as well as stress gradient terms. The VOF method is used to track the free 59 surface. The transport equations for k and  $\varepsilon$  are solved with the similar method used in solving the 61 momentum equations (Lin and Liu, 1998a, b).

#### 3. Numerical results and discussions

To validate the numerical model, numerical simulations of several laboratory experiments have 67 been carried out, including waves generated by vertical bottom movements (Hammack, 1973) and 69 by a sliding triangular block on a uniform beach (Heinrich, 1992). In Hammack's experiments waves 71 do not break in the generation region and the present numerical results agree with Hammack's 73 data very well. In this paper we shall focus our discussion on Heinrich's experiments in which the 75 generated waves break.

The computational domain is 12 m in x-direction 77 and 2 m in *y*-direction. A variable grid size system is used in the x-direction with minimum grid size of 79 0.01 m and a fixed grid size of 0.01 m is employed in 81 y-direction. To satisfy all stability conditions and restrictions of the incorporated methods, a fixed time step of  $5 \times 10^{-4}$  s is used. Numerical results in 83 generation (i.e., near moving mass) and propagation regions are compared with experimental data as 85 shown in Figs. 1 and 2. The submarine mass movement is modeled by a triangular shaped 87 moving boundary that is initially located at 0.01 m below the free surface as in Heinrich (1992). The 89 measured displacement time history from the Heinrich experiment is used as prescribed motion 91 of the triangular mass. Since the grid size is not small enough to resolve the boundary layer, the 93 free-slip boundary condition is applied on all the solid boundaries including sliding body, slopes, and 95 channel bottom. As shown in Figs. 1 and 2, wave profiles in the generation region and the propaga-97 tion region are in good agreement with experimental data. However, some deviations are observed in 99 wave profile at t = 1.5 s when the reflected wave starts to break. It is surmised that the disagreement 101 in wave profile is caused by the random nature of turbulence near wave breaking where the "exact" 103 measured value is difficult to determine.

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29 81 Fig. 1. Free-surface comparisons between simulation and experimental data at 0.5, 1.0, 1.5, 2.0, 2.5, and 3.0s in wave generation region. First panel shows portion of triangular shape moving boundary. 83

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33 A convergence test using minimum grid sizes of 0.005, 0.01, 0.02, and 0.04 m has been performed. A 35 fine grid of 30 cells is used to resolve maximum wave height. It is observed that convergence is achieved 37 with a grid size 0.01 m. This value (or smaller) is employed throughout the study.

39 Turbulence generation by the submarine mass movement on a beach and its evolution are 41 examined. Fig. 3 shows the contours of turbulence intensity at t = 0.5, 1.0, 1.5, 2.0, 2.5, and 3.0 s. It is

- observed that when the mass is in motion turbulence 43 is generated around the upper right corner because 45 of flow separation. Once the waves generated by the
- moving mass reach shore, waves are reflected. After
- 47 the mass movement stops, turbulence is generated by the breaking of the reflected wave near the free
- 49 surface and turbulence intensity decreases gradually. The maximum turbulence intensity can reach
- 51 0.83 m/s, which is almost 50% of the mean velocity.

The influence of the submarine mass movement 85 velocity is examined by varying the displacement time history. Denoting by  $a_0$  as the initial accelera-87 tion of the mass movement measured in the experiment, we have calculated three additional 89 cases with accelerations that are  $0.5a_0$ ,  $0.75a_0$  and  $1.25a_0$ , respectively. In these simulations the total 91 displacement and the volume of mass movement remain constant so that only one parameter, i.e., 93 velocity of the moving mass, is varied. The effects of mass movement velocity on maximum wave heights, 95 runup and rundown are shown in Figs. 4 and 5, respectively. As expected, the magnitudes of the 97 wave height, runup and rundown increase with increasing acceleration. 99

Another case examined is an aerial slide in which a part of the moving body is initially located above 101 the free surface and slides down along a uniform slope. Therefore, the moving solid boundary inter- 103 sects the free surface until the moving body is

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Fig. 2. Free-surface comparisons between simulation and experimental data at x = 4, 8, and 12 m in propagation region.

31 completely submerged. During this period of time, a special treatment in the VOF function is required to

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- 33 satisfy the law of mass conservation. This is necessary because the pressure in the free surface
  35 cell is not calculated from the Poisson pressure equation, and is specified by the free surface
- 37 boundary condition. Thus, a source/sink term cannot be used in the free surface and the moving
- 39 boundary interface cell to generate an equal amount of fluid corresponding to the volume change due to
- 41 the moving boundary. An algorithm to treat the free surface and moving-boundary interface cell is43 developed and incorporated in the code.
- Numerical simulations are performed and com-45 pared with the experimental data obtained by Heinrich (1992) to validate the predictive capability
- 47 of the numerical model for an aerial sliding. The
- problem setup is exactly the same as that for thesubmarine slide except that the moving body islocated initially just above the free surface. Wave
- 51 profiles in the generation region at t = 0.6, 1.0, and 1.5 s are compared with experimental data as shown

in Fig. 6. From the wave profile at t = 0.6 s, we 53 observe that the wave starts to break and becomes highly random. The discrepancy of wave profiles at 55 t = 1.5 s might be attributed to turbulence.

The numerical model developed in this study is 57 utilized to investigate the functional relationship between both the runup and rundown of submarine 59 slide generated waves and the geometric parameters of the sliding body and slope. From the previous 61 work by Chen (2002), the following form of functional relation is employed. 63

$$\frac{\eta_{rd}}{b} = c_0 \gamma^{c_1} \left(\frac{A_l}{A_w}\right)^{c_2} (\sin \theta)^{c_3} (\sin \beta)^{c_4}, \tag{6}$$

$$\frac{\eta_{up}}{b} = d_0 \gamma^{d_1} \left(\frac{A_l}{A_w}\right)^{d_2} (\sin\theta)^{d_3} (\sin\beta)^{d_4}.$$
(7) 69

In the above equation,  $c_0$ ,  $c_1$ ,  $c_2$ ,  $c_3$ ,  $c_4$ ,  $d_0$ ,  $d_1$ ,  $d_2$ ,  $d_3$ , 71 and  $d_4$  are constants to be determined,  $\eta_{rd}$  the maximum rundown,  $\eta_{up}$  the maximum runup, b the 73 base length of triangular sliding body,  $\theta$  the slope angle,  $\gamma$  the specific weight of sliding body,  $\beta$  the 75 angle of top face of sliding body,  $A_I$  the area of sliding body, and  $A_w$  the area of fluid above the 77 sliding body. A series of numerical experiments is conducted to examine the functional relations. 79

In previous studies (Chen, 2002; Grilli and Watts, 1999) of functional relations between submarine 81 slide and runup/rundown, the motion of sliding body is determined by solving the differential 83 equation obtained by balancing inertial, added mass, gravitational, buoyancy, and fluid dynamic 85 drag forces. In this study, the sliding body movement is not predetermined but obtained by con-87 sidering the instantaneous dynamic equilibrium of the moving body including the coupled fluid-struc-89 ture interaction. An iterative procedure is introduced to compute the sliding body movement. For 91 the parametric study of maximum runup and rundown presented in this study, no prescribed 93 sliding body motion is used because there is no experimental data available. Details of the fluid--95 structure interaction modeling will be presented in a future paper. The numerical results shown in Fig. 1 97 are obtained by using predetermined time history of sliding block. The time history of sliding block 99 measured from the experiment conducted by Heinrich (1991) is used for sliding block motion in 101 that particular simulation.

For the numerical experiments for runup and 103 rundown, the computational domain is discretized

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Fig. 4. Influence of sliding mass velocity on wave height: (a) time series of free surface at x = 3.5 m, and (b) maximum wave height.

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D. Yuk et al. / Computers & Geosciences I (IIII) III-III



11 Fig. 5. Influence of mass movement velocity on runup and rundown: (a) elevation of free surface level along solid fixed boundary, and (b) 63 maximum and minimum free surface level.



Fig. 6. Free surface comparisons between simulation and experimental data at 0.5, 1.0, and 1.5s in wave generation region. Solid rectangles shows upper right corner of triangular shape moving boundary.
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by 410 × 280 grids points in horizontal and vertical
direction, respectively, and variable time step is used
to advance solutions in time so that stability
conditions are satisfied. The slope where landslides
occur and runup/rundown is measured is located on



7

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Fig. 7. Computational domain and numerical experiment setup.

the left end of computational domain. In addition to
specifying the domain boundary at the right edge as
"open", a sponge layer of sufficient width is placed
on the right side to prevent reflections of waves at
domain boundary and ensure full energy absorption
(see Fig. 7).

Four sets of numerical experiments are conducted. In each set, only one parameter is varied with all others being fixed so that the effects of varying the particular parameter can be examined. The parameter space used in this study is shown in Table 1. The range of parameter variation is determined by considering the possibility of physical realization. For example, the specific density of landslides can be less than 1.0, but physically it may not be realizable because of the buoyant force.

To measure the runup and rundown, numerical 95 wave gauges are placed along the slope. However, maximum and minimum vertical elevations of the 97 free surface on the slope are recorded as runup and rundown, respectively. The distance that waves 99 move along the slope can also be calculated using the maximum and minimum values in the vertical 101 direction and the slope angle.

Figs. 8 and 9 show the effects of parameters 103 considered in this study on rundown/runup and the

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#### D. Yuk et al. / Computers & Geosciences I (IIII) III-III

1 Table 1 Parameters used for runup and rundown simulations

-								
3	Test	$\sin \theta$	γ	β	$A_l$	$A_w$	$A_l / A_w$	$\sin\beta$
5	1	0.707	1.4	1.0	0.250	0.240	1.0399	0.707
	2	0.707	1.8	1.0	0.250	0.240	1.0399	0.707
7	3	0.707	2.0	1.0	0.250	0.240	1.0399	0.707
/	4	0.707	2.4	1.0	0.250	0.240	1.0399	0.707
	5	0.707	2.8	1.0	0.250	0.240	1.0399	0.707
9	6	0.707	2.12	0.707	0.125	0.071	1.768	0.707
	7	0.707	2.12	0.707	0.125	0.115	1.083	0.707
11	8	0.707	2.12	0.707	0.125	0.145	0.865	0.707
	9	0.707	2.12	0.707	0.125	0.180	0.695	0.707
10	10	0.707	2.12	0.707	0.125	0.212	0.589	0.707
13	11	0.707	2.12	0.707	0.125	0.248	0.505	0.707
	12	0.707	2.0	1.0	0.25	0.311	2.24	0.985
15	13	0.707	2.0	1.0	0.25	0.311	2.24	0.966
	14	0.707	2.0	1.0	0.25	0.311	2.24	0.940
17	15	0.707	2.0	1.0	0.25	0.311	2.24	0.866
1/	16	0.707	2.0	1.0	0.25	0.24	1.040	0.707
	17	0.643	2.0	1.0	0.25	0.24	1.040	0.707
19	18	0.574	2.0	1.0	0.25	0.24	1.040	0.707
	19	0.500	2.0	1.0	0.25	0.24	1.040	0.707
21	20	0.423	2.0	1.0	0.25	0.24	1.040	0.707
- 1	21	0.342	2.0	1.0	0.25	0.24	1.040	0.707



Fig. 8. Least-squares fit of rundown to numerical simulation data.



Fig. 9. Least-squares fit of runup to numerical simulation data.

results of regression analysis. The power curves used 53 to fit the data ensure that runup and rundown do not occur when any of parameters are zero. In 55 determining the final formula for runup and rundown, the power curves are used again and the 57 exponents from curve fit are multiplied to obtain the coefficients for final runup and rundown formula. 59

Based on the numerical results shown in Figs. 8 and 9, the functional relationships between runup/ 61 rundown and the parameters are found to be

$$\frac{\eta_{rd}}{b} = 0.4178\gamma^{0.1454} \left(\frac{A_l}{A_w}\right)^{0.2803} (\sin\beta)^{1.4395} (\sin\theta)^{0.5086},$$
(8)

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$$\frac{\eta_{up}}{b} = 1.0593\gamma^{0.2078} \left(\frac{A_l}{A_w}\right)^{0.1889} (\sin\beta)^{3.6134} (\sin\theta)^{1.6566}.$$
(9)

71 Note that larger runup and rundown are observed as expected for increasing mass density, face angle, slope angle, and decreasing initial submergence of the landslide. 75

#### 4. Concluding remarks

The capability and accuracy of the present 79 numerical model in predicting wave generation by submarine and aerial mass movements and propagation has been validated. In addition, the influence of moving body velocity on runup and rundown has 83 been examined. For the higher sliding body velocity, maximum runup and rundown are increased as 85 expected.

Turbulence generation by triangular shape moving body occurs around the upper right corner due to flow separation and near the free surface where waves break. Careful experiments measuring the velocity field are desirable to validate the prediction 91 of the turbulence intensity.

Relationships between maximum runup and 93 maximum rundown as functions of specific density, initial submergence level, angle of the moving mass 95 as well as slope angle are identified. The runup and rundown formulae show good agreement with 97 physical intuitions.

Finally we should remark that the present results 99 are limited to 2D slides, which are uniform along the shoreline. In reality slides are 3D. The predicted 101 maximum runup based on the present 2D slides might not be conservative. In the case of a 3D slide, 103 additional lateral (in the alongshore direction) as

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# ARTICLE IN PRESS

#### D. Yuk et al. / Computers & Geosciences I (IIII) III-III

- well as the on-offshore waves can be generated due to the free surface drawdown and rebound above
   the moving slide. This feature requires further study.
- 5

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