## SECTION 1: INTRODUCTION

ESE 430 - Feedback Control Systems

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Introduction

## Introduction

$\square$ In ESE 330, you learned how to model dynamic systems and simulate their responses
$\square$ Analysis - how does a given system respond

- Design (possibly) - tuning system parameters to achieve a desired response
$\square$ In ESE 430, you will learn how to design feedback control systems to improve the response of a given system in three primary areas:
- Dynamic response
$\square$ Steady-state error
$\square$ Stability


## Introduction

$\square$ In this section of notes we will take a look at a simple motor-driven rack and pinion positioning system example to do the following:

- Review dynamic system modeling fundamentals
- Bond graphs
- State-variable models
- System poles/zeros
- Transient response - step, impulse, ...
- Frequency response
- Introduce feedback control
- What is it?
- How can it help us obtain a desired system response?


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## Dynamic System Modeling

The design of a feedback control system requires first having a model of the system to be controlled.
This sub-section provides a review of dynamic system modeling and analysis fundamentals.

## Rack and Pinion Positioning System

$\square$ Simplified rack and pinion positioning system

- E.g., automated assembly equipment, print-head driver, etc.

- Voltage source drives a DC motor
- Motor inductance neglected here
- Motor turns shaft and pinion gear
- As pinion turns, rack translates
- The thing to be positioned is attached to the rack
- Rack connection has both compliance and damping


## Bond-Graph Model

$\square$ Create a bond-graph model
$\square$ First, annotate the schematic

- Label all node voltages

- Indicate assumed positive voltage polarities and current direction
- Label velocities at each mass and end of each spring and damper
- Choose displacements of springs and dampers to be positive in either compression or tension


## Bond-Graph Model



## Element

Voltage/Velocity
$\square$ Next, tabulate all one- and two-port elements, along with relevant voltages or velocities

- Bond orientation follows power convention
- Include TF and GY equations

| Element | Voltage/Velocity |
| :---: | :---: |
| $v_{s}(t): S_{e} \rightharpoonup$ | $v_{s}$ |
| $R_{m}: R \leftharpoonup$ | $v_{R_{m}}=v_{s}-v_{a}$ |
| $-G Y \rightharpoonup$ | $v_{a}$ |
| $\omega=1 / k_{m} \cdot v_{a}$ | $\omega$ |
| $T F \rightharpoonup$ | $\omega$ |
| $v=r \cdot \omega$ | $v$ |
| $m: I \leftharpoonup$ | $v$ |
| $b: R \leftharpoonup$ | $v$ |
| $1 / k: C \leftharpoonup$ | $v$ |

## Bond-Graph Model

$\square$ Use the table to construct a bond graph

| Element | Voltage/Velocity |
| :---: | :---: |
| $v_{s}(t): S_{e} \rightharpoonup$ | $v_{s}$ |
| $R_{m}: R \leftharpoonup$ | $v_{R_{m}}=v_{s}-v_{a}$ |
| $\rightharpoonup G Y \rightharpoonup$ | $v_{a}$ |
| $\omega=1 / k_{m} \cdot v_{a}$ | $\omega$ |


| Element | Voltage/Velocity |
| :---: | :---: |
| $-T F \rightharpoonup$ | $\omega$ |
| $=r \cdot \omega$ | $v$ |
| $m: I \leftharpoonup$ | $v$ |
| $b: R \leftharpoonup$ | $v$ |
| $1 / k: C \leftharpoonup$ | $v$ |



## Bond-Graph Model

$\square$ Create computational bond graph and assign causality

$\square I_{6}$ and $C_{8}$ both have integral causality
$\square$ Two independent energy-storage elements
$\square$ A second-order system

## Bond-Graph Model

$\square$ State variables:

$$
\mathbf{x}=\left[\begin{array}{l}
p_{6} \\
q_{8}
\end{array}\right]
$$

$\square$ Annotate the bond graph in preparation for state equation derivation

- Sources
- State variable derivatives as effort/flow on independent I's and C's
- Apply constitutive laws to annotate the other power variables



## State-Variable Model

## State-Variable System Model

$\square$ Use annotated bond graph to derive a state-variable model for the system

$$
\begin{aligned}
& \dot{\mathbf{x}}=\mathbf{A} \mathbf{x}+\mathbf{B} u \\
& y=\mathbf{C} \mathbf{x}+D u
\end{aligned}
$$

$\square$ State derivatives are linear combinations of state variables and inputs

- Output is a linear combination of states and inputs
- This is a SISO system
- Single-input, single-output
$-u, y$, and $D$ are scalars


## State Equation Derivation

$\square$ Follow causality through the bond graph to express state variable derivatives as linear combinations of states and inputs

- Start with $\dot{p}_{6}$

$$
\begin{align*}
& \dot{p}_{6}=e_{6}=e_{5}-e_{7}-e_{8} \\
& \dot{p}_{6}=\frac{1}{r} e_{4}-R_{7} f_{7}-\frac{1}{C_{8}} q_{8} \\
& \dot{p}_{6}=\frac{k_{m}}{r} f_{3}-\frac{R_{7}}{I_{6}} p_{6}-\frac{1}{C_{8}} q_{8} \tag{1}
\end{align*}
$$



$$
\begin{align*}
& f_{3}=f_{2}=\frac{1}{R_{2}} e_{2}=\frac{1}{R_{2}}\left(e_{1}(t)-e_{3}\right)=\frac{1}{R_{2}} e_{1}(t)-\frac{k_{m}}{R_{2}} f_{4} \\
& f_{3}=\frac{1}{R_{2}} e_{1}(t)-\frac{k_{m}}{R_{2}} \frac{1}{r} f_{5}=\frac{1}{R_{2}} e_{1}(t)-\frac{k_{m}}{R_{2} r} \frac{1}{I_{6}} p_{6} \tag{2}
\end{align*}
$$

## State Equation and Output Equation

$\square$ Substituting (2) into (1) give the first state equation:

$$
\begin{equation*}
\dot{p}_{6}=-\frac{k_{m}^{2}+R_{7} r^{2} R_{2}}{r^{2} R_{2} I_{6}} p_{6}-\frac{1}{C_{8}} q_{8}+\frac{k_{m}}{r R_{2}} e_{1}(t) \tag{3}
\end{equation*}
$$

$\square$ Next, move on to $\dot{q}_{8}$

$$
\dot{q}_{8}=f_{8}=f_{6}=\frac{1}{I_{6}} p_{6}
$$

$\square$ The second state equation:

$$
\begin{equation*}
\dot{q}_{8}=\frac{1}{I_{6}} p_{6} \tag{4}
\end{equation*}
$$


$\square$ The output is the position of the rack, which is also the displacement of the spring

$$
\begin{equation*}
y=q_{8} \tag{5}
\end{equation*}
$$

## State-Variable System Model

$\square$ Equations (3) - (5) can be assembled in matrix form to give the state-variable system model:

$$
\begin{align*}
& {\left[\begin{array}{l}
\dot{p}_{6} \\
\dot{q}_{8}
\end{array}\right]=\left[\begin{array}{cc}
-\frac{k_{m}^{2}+R_{7} r^{2} R_{2}}{r^{2} R_{2} I_{6}} & -\frac{1}{C_{8}} \\
\frac{1}{I_{6}} & 0
\end{array}\right]\left[\begin{array}{c}
p_{6} \\
q_{8}
\end{array}\right]+\left[\begin{array}{c}
\frac{k_{m}}{r R_{2}} \\
0
\end{array}\right] e_{1}(t)}  \tag{6}\\
& y=\left[\begin{array}{ll}
0 & 1
\end{array}\right]\left[\begin{array}{l}
p_{6} \\
q_{8}
\end{array}\right]
\end{align*}
$$

$\square$ Substituting in physical system parameters:

$$
\begin{align*}
& {\left[\begin{array}{l}
\dot{p} \\
\dot{x}
\end{array}\right]=\left[\begin{array}{cc}
-\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} & -k \\
\frac{1}{m} & 0
\end{array}\right]\left[\begin{array}{l}
p \\
x
\end{array}\right]+\left[\begin{array}{c}
\frac{k_{m}}{r R_{m}} \\
0
\end{array}\right] e_{1}(t)}  \tag{7}\\
& y=\left[\begin{array}{ll}
0 & 1
\end{array}\right]\left[\begin{array}{l}
p \\
x
\end{array}\right]
\end{align*}
$$

## Transfer Function

## Transfer Function

$\square$ The state-variable model is one of many possible mathematical models for the system

- A time-domain model
$\square$ The transfer function is another model
- A Laplace-domain model
$\square$ The ratio of the output to the input in the Laplace domain, assuming zero initial conditions:

$$
G(s)=\frac{Y(s)}{U(s)}
$$

- Useful for determining the Laplace-domain output

$$
Y(s)=G(s) \cdot U(s)
$$

- System poles/zeros are readily apparent
- Substitute $s \rightarrow j \omega$ for frequency response function


## Transfer Function

$\square$ A couple of ways to convert from the state-variable model to the transfer function

- Calculate directly:

$$
G(s)=\mathbf{C}(s \mathbf{I}-\mathbf{A})^{-1} \mathbf{B}+\mathbf{D}
$$

- Requires matrix inversion
$\square$ Solve for the states using Cramer's rule, combine according to the output equation and solve algebraically for $G(s)=Y(s) / U(s)$
- Matrix inversion is not required
$\square$ We'll step through both methods


## State Space $\rightarrow$ Transfer Function - 1

$\square$ From the ABCD matrices that define the state variable model:

$$
G(s)=\mathbf{C}(s \mathbf{I}-\mathbf{A})^{-1} \mathbf{B}+\mathbf{D}
$$

$\square$ For our example:

$$
G(s)=\left[\begin{array}{ll}
0 & 1
\end{array}\right]\left[\begin{array}{cc}
s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} & k \\
-\frac{1}{m} & s
\end{array}\right]^{-1}\left[\begin{array}{c}
\frac{k_{m}}{r_{m}} \\
0
\end{array}\right]
$$

$\square$ The inverse of the $(s I-A)$ matrix is

$$
(s \mathbf{I}-\mathbf{A})^{-1}=\frac{\operatorname{adj}(\mathbf{s} \mathbf{I}-\mathbf{A})}{|s \mathbf{I}-\mathbf{A}|}=\frac{\operatorname{adj}(\mathbf{s I}-\mathbf{A})}{\Delta(s)}
$$

$\square \Delta(s)$ is the characteristic polynomial of the system

## The Characteristic Polynomial

$\square$ The characteristic polynomial

$$
\begin{aligned}
& \Delta(s)=|s I-A|=\left|\begin{array}{cc}
s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} & k \\
-\frac{1}{m} & s
\end{array}\right| \\
& \Delta(s)=s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}
\end{aligned}
$$

$\square$ Recall
$\square \Delta(s)$ is the denominator of the Laplace transform of every state and the output
$\square$ The roots of $\Delta(s)$, the poles of the system, determine the nature of the system response

## State Space $\rightarrow$ Transfer Function - 1

$$
(s \mathbf{I}-\mathbf{A})^{-1}=\frac{1}{\Delta(s)}\left[\begin{array}{cc}
s & -k \\
\frac{1}{m} & s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m}
\end{array}\right]
$$

$\square$ Substituting back into the expression for $G(s)$

$$
\left.\begin{array}{c}
G(s)=\frac{1}{\Delta(s)}\left[\begin{array}{ll}
0 & 1
\end{array}\right]\left[\begin{array}{cc}
s & -k \\
\frac{1}{m} & s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m}
\end{array}\right]\left[\begin{array}{c}
\frac{k_{m}}{r R_{m}} \\
0
\end{array}\right] \\
G(s)=\frac{1}{\Delta(s)}\left[\frac{1}{m} \quad s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m}\right.
\end{array}\right]\left[\begin{array}{c}
\frac{k_{m}}{r R_{m}} \\
0
\end{array}\right] \quad .
$$

$\square$ The system transfer function:

$$
G(s)=\frac{k_{m} / m R_{m} r}{s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}}
$$

## State Space $\rightarrow$ Transfer Function - 2

$\square$ Alternatively, find $G(s)$ by Laplace transforming the state equation and applying Cramer's rule
$\square$ The state equation in general form:

$$
\begin{equation*}
\dot{\mathbf{x}}=\mathbf{A x}+\mathbf{B u} \tag{1}
\end{equation*}
$$

$\square$ Apply the Laplace transform, assuming zero initial conditions

$$
s \mathbf{X}(s)=\mathbf{A} \mathbf{X}(s)+\mathbf{B} U(s)
$$

$\square$ Collecting the transform of the state vector on the left-hand side

$$
s \mathbf{X}(s)-\mathbf{A X}(s)=\mathbf{B} U(s)
$$

$\square$ Factoring out $\mathbf{X}(s)$

$$
\begin{equation*}
(s \mathbf{I}-\mathbf{A}) \mathbf{X}(s)=\mathbf{B} U(s) \tag{2}
\end{equation*}
$$

## State Space $\rightarrow$ Transfer Function - 2

$\square$ Can now apply Cramer's rule to solve for individual elements in $\mathbf{X}(s)$, i.e., the Laplace transform of individual states

$$
X_{i}(s)=\frac{\left|(s \mathbf{I}-\mathbf{A})_{i}\right|}{|s \mathbf{I}-\mathbf{A}|}
$$

$\square(s \mathbf{I}-\mathbf{A})_{i}$ is the matrix formed by replacing the $i^{\text {th }}$ column of ( $s \mathbf{I}-\mathbf{A}$ ) with $\mathbf{B} U(s)$, the RHS of (2)
$\square$ Determine as many states as are required to calculate $Y(s)$

## State Space $\rightarrow$ Transfer Function - 2

$\square$ State variable model for our system

$$
\begin{aligned}
& {\left[\begin{array}{l}
\dot{p} \\
\dot{x}
\end{array}\right]=\left[\begin{array}{cc}
-\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} & -k \\
\frac{1}{m} & 0
\end{array}\right]\left[\begin{array}{c}
p \\
x
\end{array}\right]+\left[\begin{array}{c}
\frac{k_{m}}{r R_{m}} \\
0
\end{array}\right] e_{1}(t)} \\
& y=\left[\begin{array}{ll}
0 & 1
\end{array}\right]\left[\begin{array}{l}
p \\
x
\end{array}\right]
\end{aligned}
$$

$\square$ Here the output depends only on the displacement of the spring, $y(t)=x(t)$

- To find $Y(s)$, apply Cramer's rule to the Laplace transformed state equation to find $X(s)$
- NOTE: $X(s)$ is the Laplace transform of the displacement of the spring, $\mathbf{X}(s)$ is the Laplace transform of the state vector


## State Space $\rightarrow$ Transfer Function - 2

$$
X(s)=\frac{\left|(s \mathbf{I}-\mathbf{A})_{2}\right|}{|s \mathbf{I}-\mathbf{A}|}=\frac{1}{\Delta(s)}\left|\begin{array}{cc}
s+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} & \frac{k_{m}}{R_{m} r} U(s) \\
-\frac{1}{m} & 0
\end{array}\right|=\frac{k_{m} / m R_{m} r U(s)}{s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}}
$$

$\square$ The Laplace transform of the output is

$$
Y(s)=X(s)=\frac{k_{m} / m R_{m} r U(s)}{s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}}
$$

$\square$ Dividing both sides by the input gives the transfer function

$$
G(s)=\frac{Y(s)}{U(s)}=\frac{k_{m} / m R_{m} r}{s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}}
$$

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## System Poles \& Zeros

## System Poles \& Zeros

$$
G(s)=\frac{k_{m} / m R_{m} r}{s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}}
$$

$\square$ Poles: values of $s$ for which $G(s)=\infty$
$\square$ The roots of the denominator, $\Delta(s)$
$\square$ Solutions to the characteristic equation, $\Delta(s)=0$

- Here, there are two poles
$\square$ Zeros: values of $s$ for which $G(s)=0$
$\square$ The roots of the numerator polynomial
$\square$ Here, there are none


## Natural Frequency \& Damping Ratio

$\square$ This second-order characteristic polynomial

$$
\Delta(s)=s^{2}+\frac{k_{m}^{2}+b r^{2} R_{m}}{r^{2} R_{m} m} s+\frac{k}{m}
$$

can be re-written as

$$
\Delta(s)=s^{2}+2 \zeta \omega_{n} s+\omega_{n}^{2}
$$

$\square \zeta$ is the damping ratio

$$
\zeta=\frac{k_{m}^{2}+b r^{2} R_{m}}{2 \sqrt{k m} r^{2} R_{m}}
$$

$\square \omega_{n}$ is the natural frequency

$$
\omega_{n}=\sqrt{\frac{k}{m}}
$$

## System Poles \& Zeros

$\square$ The second-order system has two poles at

$$
s_{1,2}=-\zeta \omega_{n} \pm \omega_{n} \sqrt{\zeta^{2}-1}
$$

$\square$ The value of the damping ratio, $\zeta$, determines the nature of the two poles:
$\square \zeta>1$ : two real, distinct poles - over-damped
$\square \zeta=1$ : two real, identical poles - critically-damped
$\square \zeta<1$ : complex-conjugate pair poles - under-damped
$\square$ Type of poles, and, therefore, the value of $\zeta$, determines the nature of the response

## System Poles \& Zeros

$\square$ Assume the following system parameter values:

- $R_{m}=8 \Omega$
- $k_{m}=0.02 \mathrm{Nm} / \mathrm{A}$
- $r=0.01 \mathrm{~m}$

ㅁ $m=0.1 \mathrm{~kg}$

- $k=0.5 \mathrm{~N} / \mathrm{m}$
- $b=0.05 \mathrm{Ns} / \mathrm{m}$
$\square$ Poles:
- $s_{1}=-1.15 \mathrm{rad} / \mathrm{sec}$
- $s_{2}=-4.35 \mathrm{rad} / \mathrm{sec}$

$\square \zeta=1.23>1$ - over-damped - distinct, real poles
- Monotonic step response - no overshoot or ringing


## 33 <br> Dynamic System Response

## Dynamic System Response

$\square$ Often characterize systems by their responses to particular classes of inputs, e.g.:

- Impulse response: response to an impulse with zero initial conditions - time-domain response
- Step response: response to a unit step with zero initial conditions - time-domain response
- Frequency response: system response to sinusoidal inputs of varying frequency - system gain and phase as functions of frequency - frequency-domain response
$\square$ Additionally, we often want to simulate the system's response to arbitrary inputs


## Impulse Response

$\square$ The system response in the Laplace domain is given by the product of the transfer function and the Laplace transform of the input

$$
Y(s)=G(s) \cdot U(s)
$$

$\square$ The Laplace transform of an impulse function is

$$
\mathcal{L}\{\delta(t)\}=1
$$

therefore, a system's impulse response is the inverse Laplace transform of its transfer function

$$
g(t)=\mathcal{L}^{-1}\{G(s)\}
$$

## Impulse Response

$\square$ For our rack and pinion positioning system:

$$
g(t)=\mathcal{L}^{-1}\{G(s)\}=\mathcal{L}^{-1}\left\{\frac{2.5}{s^{2}+5.5 s+5}\right\}
$$

$\square$ Inverse transform via partial fraction expansion

$$
\begin{align*}
& G(s)=\frac{2.5}{s^{2}+5.5 s+5}=\frac{r_{1}}{s+1.15}+\frac{r_{2}}{s+4.35}  \tag{1}\\
& 2.5=r_{1}(s+4.35)+r_{2}(s+1.15) \\
& 2.5=\left(r_{1}+r_{2}\right) s+4.35 r_{1}+1.15 r_{2}
\end{align*}
$$

$\square$ Equating coefficients and solving the resulting system of two equations gives the following residues:

$$
\begin{aligned}
& r_{1}=0.7809 \\
& r_{2}=-0.7809
\end{aligned}
$$

## Impulse Response

$\square$ Substituting the residues back into (1) gives

$$
G(s)=\frac{0.7809}{s+1.15}-\frac{0.7809}{s+4.35}
$$

$\square$ Inverse Laplace transforming gives the impulse response:


$$
g(t)=0.7809 e^{-1.15 t}-0.7809 e^{-4.35 t}
$$

## Step Response

$\square$ The step response in the Laplace domain is

$$
\begin{aligned}
& Y(s)=U(s) \cdot G(s) \\
& Y(s)=\frac{1}{s} \cdot \frac{2.5}{s^{2}+5.5 s+5}
\end{aligned}
$$

$\square$ Inverse transforming gives the time-domain step response:

Rack and Pinion Positioning System Step Response


$$
y(t)=0.5-0.6795 e^{-1.15 t}+0.1795 e^{-4.35 t}
$$

## Frequency Response

$\square$ The frequency response function or sinusoidal transfer function is obtained by substituting $j \omega$ for $s$ in the transfer function

$$
G(j \omega)=\frac{2.5}{(j \omega)^{2}+5.5(j \omega)+5}
$$

$\square$ This complex function of frequency can be evaluated to give the system's:

- Gain: the ratio of the magnitudes of the system's (sinusoidal) output to input as function of frequency
- Phase: the phase shift from the (sinusoidal) input to the output as a function of frequency


## Frequency Response

$\square$ Gain and phase are plotted as a Bode plot:


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## Open-Loop System Response

## Block Diagram Model

$\square$ Our positioning system, or plant, can be represented in block diagram form as

$\square$ It has an input, $u(t)$, and an output, $y(t)$

- Input/output relationship described by the plant model: transfer function, state variable model, etc.


## Open-Loop Configuration

$\square$ Say we want to command a 5 cm displacement from our plant

- Plant input, $u(t)$, is a voltage applied to a motor
$\square$ We'd like the input to be the desired displacement, e.g. 5 cm
- This desired output specified by the reference input, $r(t)$
$\square$ Block diagram is now:

$\square$ Added a controller block
- Constant gain, $K_{O L}$, to convert from $r(t)$ to $u(t)$
- Value of $K_{O L}$ depends on properties of the plant


## Open-Loop Controller Gain


$\square$ How do we determine $K_{O L}$ ?
$\square$ The steady-state gain of the system is

$$
G_{s S}=\lim _{s \rightarrow 0} G(s)=\lim _{s \rightarrow 0}\left(\frac{2.5}{s^{2}+5.5 s+5}\right)=0.5 \mathrm{~m} / V
$$

- Set $K_{O L}=1 / G_{S S}=2 \mathrm{~V} / \mathrm{m}$
$\square$ Say, for example, that we want a displacement of 5 cm

$$
\begin{aligned}
& r(t)=0.05 \mathrm{~m} \\
& u(t)=K_{O L} \cdot r(t)=2 \mathrm{~V} / \mathrm{m} \cdot 0.05 \mathrm{~m}=100 \mathrm{mV} \\
& y_{s s}=u(t) \cdot G_{s s}=100 \mathrm{mV} \cdot 0.5 \mathrm{~m} / \mathrm{V}=5 \mathrm{~cm}
\end{aligned}
$$

## Open-Loop Response

$\square$ The open-loop controller yields a steady-state output equal to the reference input

$$
y_{s s}=r(t)
$$



## Disturbance Input

$\square$ Consider what happens if some external factor affects the system

- Additional load
- Increased drag due to part wear, etc.
$\square$ This is a disturbance
$\square$ Model as an additional input to the plant, $w(t)$ :



## Effect of Disturbance

$\square$ The open-loop controller ( $K_{O L}$ ) was designed for the specific plant characteristics

- Disturbance not accounted for
$\square$ Now,

$$
\begin{aligned}
& y_{s S}=\left[K_{O L} \cdot r(t)+w(t)\right] G_{S S} \\
& y_{s s}=r(t)+w(t) \cdot G_{s s}
\end{aligned}
$$

$\square$ Steady-state error results

$$
\begin{aligned}
& y_{s s} \neq r(t) \\
& e_{s s}=r(t)-y_{s s}
\end{aligned}
$$



# Closed-Loop Feedback Control 

## Feedback Control

$\square$ Steady-state error due to disturbance input can be addressed by adding feedback
$\square$ The output is measured and fed back to the input
$\square$ Subtracted from the reference input - negative feedback


## Feedback Control


$\square$ This is a closed-loop configuration
$\square$ Difference between reference (desired output), $r(t)$, and actual output, $y(t)$, is the error, $e(t)$

$$
e(t)=r(t)-y(t)
$$

$\square$ Error gets multiplied by the closed-loop controller gain, $K_{C L}$
$\square$ Input to the plant, $u(t)$, is the controller output plus the disturbance input

$$
u(t)=K_{C L} \cdot e(t)+w(t)
$$

## Closed-Loop Response

$\square$ Because $y(t)$ is fed back and used to generate $u(t)$, error is reduced

- Though not eliminated, in this case
$\square$ Dynamics of the closed-loop system differ from the open-
 loop system


## Reducing Steady-State Error

$\square$ Increasing controller gain can further reduce steady-state error
$\square$ Closed-loop system dynamics have changed a lot
$\square$ Faster risetime, increased overshoot

$\square$ Could this pose a problem?

## Closed-Loop Poles

$\square$ Nature of open-loop and closed-loop responses differ

- Closed-loop system poles differ from openloop system poles
$\square$ Feedback moves poles

Open-Loop and Closed-Loop System Poles


## Second-Order Under-Damped System Poles

$\square$ We'll see that feedback will allow us to move poles to desirable locations
$\square$ Damping ratio
$\square$ Natural frequency
$\square$ Second-order poles:

$$
\begin{aligned}
& s_{1,2}=-\sigma \pm j \omega_{d} \\
& s_{1,2}=-\zeta \omega_{n} \pm j \omega_{n} \sqrt{1-\zeta^{2}}
\end{aligned}
$$

$$
\zeta=\frac{\sigma}{\omega_{n}}=\sin (\theta)
$$

$$
\omega_{n}=\sqrt{\sigma^{2}+\omega_{d}^{2}}
$$

$\square$ Damped natural frequency

$$
\omega_{d}=\omega_{n} \sqrt{1-\zeta^{2}}
$$

Second-Order Pole Locations


## Second-Order Under-Damped System Poles

$\square$ Pole location determines dynamic behavior
$\square$ Overshoot:

$$
\begin{aligned}
& \% O S=e^{\left(\frac{-\zeta \pi}{\sqrt{1-\zeta^{2}}}\right)} \cdot 100 \% \\
& \zeta=-\frac{\ln (O S)}{\sqrt{\pi^{2}+\ln ^{2}(O S)}}
\end{aligned}
$$

$\square$ Settling time ( $\pm 1 \%$ ) approximation:

$$
t_{s} \approx \frac{4.6}{\sigma}
$$

$\square$ Risetime approximation:

$$
t_{r} \approx \frac{1.8}{\omega_{n}}
$$

Second-Order Pole Locations


## Adding Controller Dynamics

$\square$ Previous controller was a simple gain factor

- Proportional control
$\square$ Controller could also be designed to have dynamics of its own - a compensator
- Controller transfer function may have poles and/or zeros
- Allows for better control of closed-loop system response
- Steady-state error - possible to eliminate
- Transient response - risetime, overshoot, settling time



## Closed-Loop Response

$\square$ Without getting into specifics, consider the effect of a controller that has a pole and two zeros
$\square$ Steady-state error has been eliminated
$\square$ Transient response nearly unchanged


## Closed-Loop Response

$\square$ Perhaps we want a faster response
$\square$ Alter closed-loop response by changing controller transfer function

- Much faster risetime
$\square$ Still almost no overshoot
$\square$ Still no steady-state error



## Closed-Loop Response

$\square$ Modify the controller again

- Even faster risetime
$\square$ Now, very large overshoot
$\square$ Significant ringing
- A desirable response? Perhaps not


## Controller Design


$\square$ How do we determine the controller transfer function to yield the desired response?

- The topic of this course
$\square$ What is the controller?
$\square$ A block in a block diagram? Yes.
$\square$ A mathematical function? Yes.
$\square$ But, how do we implement it?
- Electronics - digital computer or opamp circuits


## ESE 430 Course Overview

1. Introduction
2. Block Diagrams \& Signal Flow Graphs
3. Stability
4. Steady-State Error
5. Root-Locus Analysis
6. Root-Locus Design
7. Frequency-Response Analysis
8. Frequency-Response Design
