# SECTION 7: FREQUENCY-RESPONSE ANALYSIS

ESE 430 – Feedback Control Systems



### Introduction

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- We have seen how to design feedback control systems using the *root locus*
- In the final two sections of the course, we'll learn how to do the same using the open-loop *frequency response*

#### Objectives:

- Review the relationship between a system's frequency response and its transient response
- Determine static error constants from the open-loop frequency response
- Determine closed-loop stability from the open-loop frequency response

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### Relationship between Frequency Response and Transient Response

### Transient/Frequency Response Relationship

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- We have relationships some exact, some approximate
  between closed-loop pole locations and closed-loop transient response
- Also have relationships between *closed-loop frequency response* and *closed-loop transient responses*
- □ Applicable to *second-order systems*:

$$T(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

Also applicable to higher-order systems that are reasonably *approximated as second-order* Systems with a pair of dominant second-order poles

### Transient/Frequency Response Relationship

- Qualitative 2<sup>nd</sup>-order time/freq. response/pole relationships
  - Damping ratio vs. overshoot vs. peaking
  - Natural frequency vs. risetime vs. bandwidth





### **Frequency Response Peaking**

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- For systems with  $\zeta < 0.707$ , the gain response will exhibit *peaking*
- Can relate *peak magnitude* to the damping ratio

$$M_p = \frac{1}{2\zeta\sqrt{1-\zeta^2}}$$

- Relative to low-frequency gain
- And the *peak frequency* to the damping ratio and natural frequency

$$\omega_p = \omega_n \sqrt{1 - 2\zeta^2}$$



### Transient/Frequency Response Relationship

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- □ Can also relate a system's **bandwidth** (i.e., -3dB frequency,  $\omega_{BW}$ ) to the speed of its step response
- Bandwidth as a function of  $\omega_n$  and  $\zeta$

$$\omega_{BW} = \omega_n \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}}$$

Bandwidth as a function of 1% settling time and  $\zeta$ 

$$\omega_{BW} = \frac{4.6}{t_s \zeta} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}}$$

Bandwidth as a function of *peak time* and  $\zeta$ 

$$\omega_{BW} = \frac{\pi}{T_p \sqrt{1 - \zeta^2}} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}}$$



### Static Error Constants

- For unity-feedback systems, open-loop transfer function gives static error constants
  - Use static error constants to calculate steady-state error

$$K_{p} = \lim_{s \to 0} G(s)$$
$$K_{v} = \lim_{s \to 0} sG(s)$$
$$K_{a} = \lim_{s \to 0} s^{2}G(s)$$

We can also determine static error constants from a system's open-loop Bode plot

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For a type 0 system

- $K_p = \lim_{s \to 0} G(s)$
- At low frequency, i.e.
  below any open-loop
  poles or zeros

 $G(s) \approx K_p$ 

Read K<sub>p</sub> directly from
 the open-loop Bode plot
 Low-frequency gain



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For a type 1 system

$$K_{v} = \lim_{s \to 0} sG(s)$$

At low frequencies, i.e. below any other open-loop poles or zeros

$$G(s) \approx \frac{K_v}{s}$$
 and  $|G(j\omega)| \approx \frac{K_v}{\omega}$ 

- $\Box$  A straight line with a slope of  $-20 \ dB/dec$
- Evaluating this low-frequency asymptote at  $\omega = 1$  yields the velocity constant,  $K_v$
- On the Bode plot, extend the low-frequency asymptote to  $\omega = 1$

• Gain of this line at  $\omega = 1$  is  $K_v$ 





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For a type 2 system

$$K_a = \lim_{s \to 0} s^2 G(s)$$

At low frequencies, i.e. below any other open-loop poles or zeros

$$G(s) \approx \frac{K_a}{s^2}$$
 and  $|G(j\omega)| \approx \frac{K_a}{\omega^2}$ 

- $\Box$  A straight line with a slope of  $-40 \ dB/dec$
- Evaluating this low-frequency asymptote at  $\omega = 1$  yields the acceleration constant,  $K_a$
- On the Bode plot, extend the low-frequency asymptote to  $\omega = 1$ 
  - **Gain of this line at**  $\omega = 1$  is  $K_a$



# <sup>16</sup> Frequency Response & Stability

### Stability

Consider the following system



- We already have a couple of tools for assessing stability as a function of loop gain, K
  - Routh Hurwitz
  - Root locus
- Root locus:

Stable for some values of KUnstable for others



# Stability

- In this case gain is stable
  *below* some value
- Other systems may be stable for gain *above* some value
- Marginal stability point:
  Closed-loop poles on the imaginary axis at ±jω<sub>1</sub>
  For gain K = K<sub>1</sub>



### **Open-Loop Frequency Response & Stability**

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- Marginal stability point occurs when closed-loop poles are on the imaginary axis
  - **\square** Angle criterion satisfied at  $\pm j\omega_1$

 $|KG(j\omega_1)| = 1$  and  $\angle KG(j\omega_1) = -180^{\circ}$ 

• Note that  $-180^\circ = 180^\circ$ 

*KG*(*j*ω) is the *open-loop frequency response Marginal stability* occurs when:
 Open-loop gain is: *KG*(*j*ω) = 0 *dB* Open-loop phase is: ∠*KG*(*j*ω) = -180°

# Stability from Bode Plots

- Varying K simply shifts gain response up or down
- Here, stable for smaller gain values
  - $|KG(j\omega)| < 0 \ dB$  when  $\angle KG(j\omega) = -180^{\circ}$
- Often, stable for larger gain values
  - $\square |KG(j\omega)| > 0 \, dB \text{ when}$ 
    - $\angle KG(j\omega) = -180^{\circ}$
- Root locus provides this information
  Bode plot does not



### **Open-Loop Frequency Response & Stability**

A method does exist for determining stability from the open-loop frequency response:

#### Nyquist stability criterion

- Graphical technique
- Uses open-loop frequency response
- Determine system stability
- Determine gain ranges for stability
- Before introducing the Nyquist criterion, we must first introduce the concept of *complex functional mapping*



# **Complex Functional Mapping**

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Consider a complex function

$$F(s) = \frac{(s - z_1)(s - z_2) \cdots}{(s - p_1)(s - p_2) \cdots}$$

Takes one complex value, s, and yields a second complex value, F(s)

**I** In other words, it **maps** s to F(s)



# Mapping of Contours

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- □ *F*(*s*) provides a mapping of individual points in the s-plane to corresponding points in the F-plane
- Can also map all points around a *contour* in the splane to another contour in the F-plane



# Mapping of Contours

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- Recall how we approached the application of the angle criterion
  - Vector approach to the evaluation of a transfer function at a particular point in the s-plane

$$|G(s_1)| = \frac{\prod |vectors\ from\ zeros\ to\ s_1|}{\prod |vectors\ from\ poles\ to\ s_1|}$$

 $\angle G(s_1) = \Sigma \angle (from \ zeros \ to \ s_1) - \Sigma \angle (from \ poles \ to \ s_1)$ 

 Can take the same approach to evaluating complex functions around *contours* in the s-plane

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- □ Map contour A by F(s) = (s z<sub>1</sub>) in a *clockwise* direction
  □ Contour A does not enclose the zero
- $\Box$  Here, R = V, so |R| = |V| and  $\angle R = \angle V$



- □ As F(s) is evaluated around A,  $\angle V$  never exceeds 0° or 180°
- $\square$  *R* does the same:
  - Does not rotate through a full 360°
  - **Contour B does not encircle the origin**

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- Map contour A by F(s) in a *clockwise* direction
  Contour A does not enclose the pole
- □ Here, R = 1/V, so |R| = 1/|V| and  $\angle R = -\angle V$



- ∠V oscillates over some range well within 0° and 180°
  *R* rotates through the *negative* of the same range
  - Contour B does not encircle the origin

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Now, contour A encloses a single zero

$$R = V, \text{ so } |R| = |V| \text{ and } \angle R = \angle V$$



- $\Box$  V rotates through a full 360° in a clockwise direction
- $\square$  *R* does the same:
  - **Contour B encircles the origin in a** *clockwise direction*

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- Now, contour A encloses a single pole
- $\square$  R = 1/V, so |R| = 1/|V| and  $\angle R = -\angle V$



- $\Box$  V rotates through a full 360° in a clockwise direction
  - **R** rotates in the *opposite direction*
  - **Contour B encircles the origin in a** <u>CCW</u> direction

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Now, contour A encloses two poles

$$\square R = \frac{1}{V_1 V_2}, \text{ so } |R| = \frac{1}{|V_1||V_2|} \text{ and } \angle R = -(\angle V_1 + \angle V_2)$$



V<sub>1</sub> and V<sub>2</sub> each rotate through a full 360° in a clockwise direction
 R rotates in the opposite direction

**Contour B encircles the origin** twice in a <u>CCW</u> direction

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Now, contour A encloses one pole and one zero

$$\square R = \frac{V_1}{V_2}, \text{ so } |R| = \frac{|V_1|}{|V_2|} \text{ and } \angle R = \angle V_1 - \angle V_2$$



- $\Box \ \angle V_1$  and  $\angle V_2$  rotate through 360° in a CW direction
  - Their contributions rotate in opposite directions
  - $\angle R$  does not rotate through a full 360°
  - **Contour B does not encircle the origin**

### **Complex Functional Mapping of Contours**

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- Some observations regarding complex mapping of contour A in a CW direction to contour B:
  - If A does not enclose any poles or zeros, B does not encircle the origin
  - If A encloses a single pole, B will encircle the origin once in a CCW direction
  - If A encloses two poles, B will make two CCW encirclements of the origin
  - If A encloses a pole and a zero, B will not encircle the origin

#### Next, we'll use these observations to help derive the *Nyquist stability criterion*

# <sup>33</sup> Nyquist Stability Criterion

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- Our goal is to assess closed-loop stability
  Determine if there are any closed-loop poles in the RHP
- Consider a generic feedback system:



Closed-loop transfer function

$$T(s) = \frac{G(s)}{1 + G(s)H(s)}$$

Closed-loop poles are roots (zeros) of the closed-loop characteristic polynomial:

$$1 + G(s)H(s)$$

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Can represent the individual transfer functions as

$$G(s) = \frac{N_G}{D_G}$$
 and  $H(s) = \frac{N_H}{D_H}$ 

The closed-loop characteristic polynomial becomes

$$1 + G(s)H(s) = 1 + \frac{N_G}{D_G}\frac{N_H}{D_H} = \frac{D_G D_H + N_G N_H}{D_G D_H}$$

From this, we can see that:

- The **poles** of 1 + G(s)H(s) are the poles of G(s)H(s), the **open-loop poles**
- The *zeros* of 1 + G(s)H(s) are the poles of T(s), the *closed-loop poles*

- To determine stability, look for RHP closed-loop poles
- □ Evaluate 1 + G(s)H(s) CW around a contour that encircles the *entire right half-plane* 
  - Evaluate 1 + G(s)H(s) along entire *jω-axis*
  - Encircle the entire RHP with an infinite-radius arc
- If 1 + G(s)H(s) has one RHP pole, resulting contour will encircle the origin once CCW
- If 1 + G(s)H(s) has one RHP zero, resulting contour will encircle the origin once CW



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Total number of CW encirclements of the origin, N, by the resulting contour will be

$$N = Z - P$$

- P = # of RHP poles of 1 + G(s)H(s)■ Z = # of RHP zeros of 1 + G(s)H(s)
- □ Want to detect RHP **poles** of T(s), **zeros** of 1 + G(s)H(s), so

$$Z = N + P$$

- **\Box** Z = # of closed-loop RHP poles
- P = # of open-loop RHP poles
- N = # of CW encirclements of the origin



- Basis for detecting closed-loop RHP poles
  - Map contour encircling the entire RHP through closedloop characteristic polynomial
  - Count number of CW encirclements of the origin by resulting contour
  - Calculate the number of closed-loop RHP poles:

$$Z = N + P$$

- Need to know:
  - Closed-loop characteristic polynomial
  - Number of RHP poles of closed-loop characteristic polynomial

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- Instead, map through G(s)H(s)
  - Open-loop transfer function
  - Easy to use for mapping we know poles and zeros
  - Resulting contour shifts left by 1 – that's all
- □ Now, count encirclements of the point s = -1



#### Nyquist stability criterion

If a contour that encloses the entire RHP is mapped through the open-loop transfer function, G(s)H(s), then the number of closed-loop RHP poles, Z, is given by

$$Z = N + P$$

where

N = # of CW encirclements of -1P = # of open-loop RHP poles

#### Want to detect *net clockwise encirclements*

N =# CW encirclements - # CCW encirclements

- Draw a line from
  s = -1 in any
  direction
- Count number of times contour crosses the line in each direction



# 42 Nyquist Diagrams

# Nyquist Diagram

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- The contour that results from mapping the perimeter of the entire RHP is a Nyquist diagram
- Consider four segments of the contour:
  - 1) Along positive  $j\omega$ -axis, we're evaluating  $G(j\omega)H(j\omega)$ 
    - Open-loop frequency response
  - 2) Here,  $s \rightarrow C^{\infty}$ 
    - Maps to zero for any physical system
  - 3) Here, evaluating  $G(-j\omega)H(-j\omega)$ 
    - Complex conjugate of segment ①
    - Mirror ① about the real axis
  - 4) The origin
    - Sometimes a special case more later



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- Apply the Nyquist criterion to determine stability for the following system



- First evaluate along segment ①, +jω-axis
  - This is the frequency response
  - Read values off of the Bode plot



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- Segment ① is a polar plot of the frequency response
- $\square$  All of segment (2), arc at  $C^{\infty}$ , maps to the origin



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Segment ③ is the complex conjugate of segment ①
 Mirror about the real axis



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- □ Count CW encirclements of s = −1
  □ Draw a line from s = −1 in any direction
- $\Box$  Here, N = 2
- Closed-loop RHP poles given by:

$$Z = N + P$$

- □ No open-loop RHP poles, so P = 0
  - Z = 2 + 0 = 2
- Two RHP poles, so system is unstable





- This system is open-loop stable
  - Stable for low enough *K*
  - **D** Nyquist plot will not encircle s = -1
- □ Three poles and no zeros
  □ Unstable for *K* above some value
  □ Nyquist plot will encircle s = -1

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- □ For K = 30, N = 0, and the system is stable
- Modifying K simply scales the magnitude of the Nyquist plot



- Here, the Nyquist plot crosses the negative real axis at s = -0.5
- As gain increases realaxis crossing moves to the left
- Increasing K by 2x or more results in two encirclements of s =- 1
  - Unstable for K > 60More later ...
- mag. -0.2 -0.4 -0.6 -0.8



# <sup>51</sup> Poles at the Origin

# Nyquist Diagram – Poles at the Origin

- We evaluate the open-loop transfer function along a contour including the *jω*-axis
- G(jω) is undefined at the pole
  Must detour around the pole
- Consider the common case of a pole at the origin



## Nyquist Diagram – Poles at the Origin

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- Segment (4) contour:  $s = \rho e^{j\theta}$  for  $0^{\circ} \le \theta \le 90^{\circ}$
- Evaluate G(s) around segment (4) as  $\rho \rightarrow 0$

$$G(s)\Big|_{s=\rho e^{j\theta}} = \frac{1}{\rho e^{j\theta}(\rho e^{j\theta} + 2)}$$

Magnitude:

$$\left|G\left(\rho e^{j\theta}\right)\right| = \frac{1}{\rho |\rho e^{j\theta} + 2|} = \frac{1}{2\rho}$$

 $\square \operatorname{As} \rho \to 0$  $\lim_{\rho \to 0} |G(\rho e^{j\theta})| = \infty$ 

 $\square$  Maps to an arc at  $C^{\infty}$ 



# Nyquist Diagram – Poles at the Origin

### Segment ④ traversed in a CCW direction

- $\Box \theta$  varies from 0° ... + 90°
- Phase of the resulting contour:

 $\angle G\left(\rho e^{j\theta}\right) = -\theta^+$ 

- Negative because it is angle from a pole
- Extra phase from additional pole
- $\Box$  G(s) maps segment (4) to:
  - An arc at  $C^{\infty}$
  - **\square** Rotating CW from 0° to  $-90^{\circ+}$



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- Apply the Nyquist criterion to determine stability for the following system



- Use Bode plot to map segment ①
  - Infinite DC gain
  - Starts at  $-90^{\circ}$  at  $C^{\infty}$ for  $\omega = 0$



- $\Box$  Segment (1) starts at  $C^{\infty}$  at  $-90^{\circ}$
- □ Heads to the origin at −180°
- $\square$  All of segment (2), arc at  $C^{\infty}$ , maps to the origin



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# Segment ③ is the complex conjugate of segment ① Mirror about the real axis



- o Segment 4 maps to a CW arc at  $\mathcal{C}^{\infty}$ 
  - **\square** *CW*, so it does not encircle -1
  - Can't draw to scale
- $\Box$  Here, N = 0
- □ No open-loop RHP poles, so P = 0

Z = 0

No RHP poles, so system is stable



# 59 Stability Margins

# **Stability Margins**

Recall a previous example



- According to the Nyquist plot, the system is stable
  How stable?
- Two stability metrics
  - Both are measures of how close the Nyquist plot is to encircling the point s = -1
  - **Gain margin** and **phase margin**



### **Crossover Frequencies**

#### Two important frequencies when assessing stability:

#### Gain crossover frequency

The frequency at which the open-loop gain crosses 0 dB

#### Phase crossover frequency

■ The frequency at which the open-loop phase crosses −180°



# Gain Margin

- An open-loop-stable system will be closed-loop stable as long as its gain is less than unity at the phase crossover frequency
- Gain margin, GM
  - The change in openloop gain at the phase crossover frequency required to make the closed-loop system unstable



### Phase Margin

An open-loop-stable system will be closed-loop stable as long as its phase has not fallen below —180° at the gain crossover frequency

#### Phase margin, PM

The change in openloop phase at the gain crossover frequency required to make the closed-loop system unstable



### Gain and Phase Margins from Bode Plots



# Phase Margin and Damping Ratio, $\zeta$

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□ PM can be expressed as a function of damping ratio,  $\zeta$ , as

$$PM = \tan^{-1}\left(\frac{2\zeta}{\sqrt{-2\zeta^2 + \sqrt{1 + 4\zeta^4}}}\right)$$

□ For  $PM \le 65^\circ$  or so, we can approximate:

$$PM \approx 100\zeta$$
 or  $\zeta \approx \frac{PM}{100}$ 





### bode.m

### [mag,phase] = bode(sys,w)

- sys: system model state-space, transfer function, or other
- w: *optional* frequency vector in rad/sec
- mag: system gain response vector
- phase: system phase response vector in degrees
- If no outputs are specified, bode response is automatically plotted – preferable to plot yourself
- Frequency vector input is optional
  If not specified, MATLAB will generate automatically
- May need to do: squeeze (mag) and squeeze (phase) to eliminate singleton dimensions of output matrices

### nyquist.m

- sys: system model state-space, transfer function, or other
  w: optional frequency vector in rad/sec
- MATLAB generates a Nyquist plot automatically
  Can also specify outputs, if desired:

[Re,Im] = nyquist(sys,w)

Plot is not be generated in this case

### margin.m

### [GM, PM, wgm, wpm] = margin(sys)

- sys: system model state-space, transfer function, or other
- **G**M: gain margin
- PM: phase margin in degrees
- wgm: frequency at which GM is measured, the phase crossover frequency – in rad/sec
- wpm: frequency at which PM is measured, the gain crossover frequency
- If no outputs are specified, a Bode plot with GM and PM indicated is automatically generated